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UAV "Built-in" Safety Protection: A Knowledge-Centered Approach

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Presentation Plan



Problem: UAV flight safety performance prediction and protection in complex (multifactor) situations Solution approach: 'Knowledge is Power'

Methodology conceptual framework (introduction):

- micro- and macro-structural knowledge models of flight
- flight situation scenario
- operational factor
- operational hypothesis
- situational tree
- safety spectrum
- flight safety [performance] window
- 'last chance for recovery' point, self-preservation decision making
- safety chances distribution time-history
- dynamic safety window tree

Case study: 'Notional UAV Low-Altitude Flight in the Presence of Urban Infra-Structure Obstacles'

Conclusions

Natural Tree Analogy of Pilot's Situational 'Knowledge Base'



Legend:

	Characteristic zone of a pilot's situational knowledge base	Natural tree analogy
Ω	Space of possible complex flight situation scenarios	Space available for tree growth
0	Basic (standard/non-standard) flight situation scenario	Tree's trunk
1	One-factor non-standard flight situation scenario	First-order derivative branch
2	Two-factor non-standard flight situation scenario	Second-order derivative branch
A	Missing knowledge	Absent but possible branching
В	Forgotten or shadowed knowledge	Dry or broken branches
С	Non-systematic , occasionally developed knowledge	Excessive, chaotic branching
D	Fragmentary, incomplete knowledge	Insufficient, sparse branching
E	Systematic, yet economically developed and stored, knowledge	Optimally dense branching
F	Physically unattainable flight situation scenarios	A sub-domain where branching is impossible

Strengths to model by means of AI in UAVs

A, B,C, D – main defect types of a human pilot's situational knowledge.

Defects to back up by means of AI in UAVs

→ Lack of theoretical and practical training (design and testing) – especially under complex (multifactor) conditions – may result in structural disparity of a human pilot's (automaton's) internal 'situational tree' of flight.

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desirable maturity levels of AI knowledge for flight safety protection in UAVs

<u>Legend</u>: Characteristic levels of piloting expertise: $k \in \{1, 2, 3\}$ – experience of a student pilot, $k \in \{8, 9, 10\}$ – experience of a professional pilot, ace, or test pilot, $k \in \{4, ..., 7\}$ – interim (immature) states of experience.

→ The most valuable asset of an expert pilot (a perfect automaton) is the reliability and comprehensiveness of his/her (its) knowledge of the system behavior under complex (multifactor, non-standard) operational conditions. This expertise is of critical importance for reliable prediction, timely avoidance or/ and safe resolution of 'chain reaction' type emergencies in UAV flight.



Micro- and Macro- Structural Models Of Complex Flight Situation Domain



Event Process

Legend:

 \mathbf{E}_i - flight event; $\mathbf{\Pi}_i$ - flight process; \mathbf{C}_m – fuzzy constraint of flight; • - reference state; • - "bud" type state; \blacklozenge - target state ("leaf"); \land source state ("root"); **B**₋₁ – parent branch; \mathbf{B}_0 - main branch ("trunk") – basic flight scenario; $\mathbf{B}_n - n$ -th order derivative branch (non-standard scenario with *n* factors, n = 1, 2, ...

→ Micro- and macro- structures of flight are two interconnected components of the developed generalized knowledge model of a complex flight situation domain.



Basic Flight Situation Scenario (Examples)

Si	Content Description						
S ₁	Normal takeoff, maintaining commanded flight path and bank angles during initial climb						
S ₂	Normal takeoff under crosswind and given runway's surface conditions, maintaining commanded flight path and bank angles during initial climb						
S ₃	Continued takeoff (left-hand engine out at given V_{EF}), maintaining commanded flight path and bank angles during initial climb						
S 4	Normal takeoff under wind shear conditions, maintaining commanded flight path and bank angles during initial climb						
\mathbf{S}_5	Continued takeoff (left-hand engine out at V _{EF}), under crosswind conditions, maintaining commanded flight path and bank angles during initial climb						
S 6	Low-altitude level flight						

Scenario #6 will be used in the notional case study

→ Basic (Baseline) Scenario S_i is a plan of some 'central' or reference flight situation – be it standard or non-standard one. It represents the situational tree's trunk. Variations of the basic scenario – derivative cases – constitute the situational tree's crown. The vehicle's flight safety knowledge base is in fact a collection (a 'forest') of the situational trees, which are constructed for various basic scenarios and exemplify a complex (multi-factor) flight situation domain.



→ A flight situation scenario is depicted as a directed graph. It defines logic and content of flight. Scenario graph is clear and concise formal description of a flight situation. Basic scenario examples $S_1, ..., S_6$ are structurally close. They can be modified by adding new events/processes or by modifying existing ones.



Composition of Situation Scenario (S) and Operational Hypothesis (Γ) is A Situational Tree (S·Γ)





Safety Palette. Fuzzy Constraint

Safety Palette green ('norm'), ξ_G yellow/ amber ('attention'), ξ_Y red ('danger'), ξ_R black ('catastrophe'), ξ_B grey/white ('uncertainty'), ξ_W

→ Color is natural and, perhaps, the most effective and economic medium for communicating safety-related information to/ from an operator (a pilot or automaton).



→ Operational constraints under multi-factor flight conditions are not known precisely. They are inherently 'fuzzy'. The notion of fuzzy constraint (by L.A. Zadeh) and the notion of safety palette are employed for approximate measurement of the compatibility of current (i.e. measured at time instants t) system states with operational constraints for key system variables (monitored flight parameters).



Partial and Integral Safety Spectra



Integral Safety Spectrum Calculation Algorithm:

 $(\forall t) \ (t \in [t_*; t^*]) \ (\exists \xi(x_k(t)) \ (\xi(x_k(t)) \in \{\xi_W, \xi_G, \xi_Y, \xi_R, \xi_B, \ldots\} \land (\xi_W < \xi_G < \xi_Y < \xi_R < \xi_B))$ $(\xi(t) = max \ \xi(x_k(t)), \ k = 1, \ldots, p) \Longrightarrow (\xi(t) \in \Sigma \land \Sigma = \xi(t_*) \parallel \xi(t_* + \Delta) \parallel \xi(t_* + 2\Delta) \parallel \ldots \parallel \xi(t^*))$

→ For each flight situation from the situational tree, safety levels are measured for all monitored variables x_k at all recorded time instants. As a result, for each situation from the tree, a family of Partial Safety Spectra Σ_k , k = 1, ..., p, and an Integral Safety Spectrum Σ are obtained. The integral safety spectrum is a color-coded time-history of violation and restoration of monitored fuzzy constrains during a flight situation.



Safety Classification Categories

	Flight Sa	fety Category	Situation Classification Criterion					
Color	Code	Name						
	I	Safe	The system state resides mainly inside the 'green' zone. As a maximum, the system state may stay, for a <i>short time</i> , in close proximity to operational constraints, i.e. inside the 'yellow' zone, but must leave it by the end of the flight situation					
	ll-a	Conditionally Safe – a	As a maximum, the system state may stay for a <i>medium time</i> in close proximity to operational constraints, i.e. inside the 'yellow' zone					
	ll-b	Conditionally Safe – b	As a maximum, the system state may stay for a <i>long time</i> in close proximity to operational constraints, i.e. inside the 'yellow' zone					
	Ш	Potentially Unsafe	As a maximum, the system state may violate operational constraints, i.e. enter the 'red' zone, for a <i>short or medium time</i> , but must leave it by the end of the situation					
	IV	Dangerous (Prohibited)	As a maximum, the system state may stay beyond operational constraints, i.e. inside the 'red' zone, for a <i>long time</i> or till the end of the flight situation					
	V	Catastrophic ('Chain Reaction')	There is at least one (for a <i>short time</i>) occurrence of a 'black' violation of any operational constraint					

→ One more level of flight safety knowledge generalization is introduced. The goal is to measure the vehicle's safety performance in a flight situation <u>as a whole</u>. With this aim, a generalized 'safety ruler' consisting of five Safety Classification Categories I, ..., V is employed. Why five? – It is because experts cannot reliably recognize and use more than 5-10 gradations of a complex, difficult-to-formalize system-level property (e.g.: Cooper-Harper scale). New 'light green' ('salad green') and 'orange' colors have been added to the existing Safety Palette in order to denote interim Categories II-a and III, respectively.

Safety Window for Situational Tree S₁· Γ_{11} : Takeoff. Errors of Selecting Commanded Flight Path and Bank Angles in Climb



Safety Chances Distribution Pie Chart



→ Let us map safety classification levels (categories) obtained for all situations for tree $S_1 \cdot \Gamma_{11}$ onto a two-factor plane. This gives a Flight Safety [Performance] Window (FSW). In FSW above, cell **C** is located at 'column **A** - row **B**' crossing. This cell depicts safety status of <u>one</u> flight pathbranch from the tree. It is a non-standard situation with values of 30° and 14° of factors Φ_7 and Φ_{11} in S_1 . This cell is painted using the situation's Flight Safety Category color ('orange'). The FSW has a dangerous 'corner' (upper-left). Rapid transition (3) from safe ('salad green') to dangerous ('red') zone is possible (Cat. II-a \rightarrow IV), bypassing interim zones (II-b, III). Control at such 'corners' therefore requires enhanced

Category	ξj	n ^j	χ ^j , %
- I		37	28
II-a		8	6
ll-b		29	22
		1	1
IV		55	43
V		0	0
Σ <i>n</i> ^j , Σχ ^j S	$\Gamma_1 \cdot \Gamma_{11}$	130	100



Flight Safety 'Topology' Map



➔ In general, the following characteristic objects can be defined within Flight Safety 'Topology' Map:

- 1 'Abyss' (catastrophe)
- 2 'Hill' (danger)
- **3** '**Slope**' (reversible state transitions)
- 4 'Valley' (standard safety, norm)
- **5** 'Lake' (maximum safety, optimum)
- 6 'Precipice' (abrupt, irreversible state transitions, 'chain reaction')

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S₄·Γ₁₂: Normal Takeoff. 'Strong' Wind Shear. Errors of Selecting Commanded Flight Path and Bank Angles in Climb

Flight Safety Window



Safety Chances Distribution Pie Chart



→ This safety 'topology' corresponds to the tree $S_4 \cdot \Gamma_{12}$ obtained under 'strong' wind shear conditions. At small flight path angles θ_{G1} and any bank angles γ_G it reveals a stable catastrophic 'abyss' (a black strip in the bottom) and 'precipice' type transitions (6). It means that attempts of climbing at small commanded flight path angles (1° ... 2°) will inevitably lead the vehicle to a fatal outcome.

Category	ξj	n ^j	χ ^j , %
I		17	13
II-a		19	15
ll-b		19	15
		1	1
IV		61	46
V		13	10
Σ n ^j , Σχ ^j S	$_4 \cdot \Gamma_{12}$	130	100



 t_2

time

Real-Time Safety Knowledge Map (Dynamic Safety Window)

 $t = t_0$: 'benign weather' forecast



 $t = t_1$: 'strong' wind-shear warning



 $t = t_2$: 'very strong' wind-shear warning



Presented is a time-history of safety windows and safety chances distribution pie charts that correspond to a hypothetical complex flight situation domain - a union of three compositions $S_4 \cdot (\Gamma_{11} + \Gamma_{12} + \Gamma_{13})$: "Normal takeoff. Possible variations of wind-shear intensity, errors/ variations in maintaining commanded flight path and bank angles during initial climb".

→ The concept of dynamic safety window is based on use of a 'forest' of situational trees. Provided that key operational factors are measurable on board the vehicle in real time, a dynamic safety window can be used as a medium for coherent monitoring of tactical goals and constraints of flight under uncertainty.

→ Safety chances distribution pie charts are expedient to use in UAV safety indicators to monitor current state and predict the system safety chances dynamics under anticipated operational conditions during flight.

→ Note that in this particular example, the share of 'red' and 'black' scenario options increases at the expense of reducing the share of safer outcomes.

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S₆·Γ₁₄: Low-Altitude Level Flight. Errors Of Selecting Commanded Flight Path And Bank Angles

Flight Safety Window



Safety Chances Distribution Pie Chart



Category	ξj	n ^j	χ ^j , %
l l		77	33
ll-a		12	5
II-b		35	14
III		4	2
IV		82	33
V		37	15
Σ <i>n</i> ^j , Σχ ^j S	$_{6}\cdot\Gamma_{14}$	247	100

→ This Safety Window has two catastrophically dangerous 'corners' (6) corresponding to $(\theta_{G1}, \gamma_G) \cong (-10^\circ...-12^\circ, |37.5^\circ...45^\circ|)$. Sharp transition (3) of states from safe ('salad green') to dangerous ('red') zone is also possible in the left upper corner (Cat. **II-a**→**IV**), bypassing interim zones (Cat. **II-b**, **III**).





S_0 : Obstacle Approach (t_0)

Safety Chances Distribution 100 **Safety Window** \rightarrow The vehicle is approaching the obstacle 👝 nanded bank angle, deg 80 – a tower building at θ_{G2} = 0 (commanded S6. F14 -45 -37.5 -30 22.5 30 37.5 flight path angle) and γ_{G} = -15° 24 60 deg. χ^j, % 22 (commanded bank angle). No threat is Commanded flight path angle (climb, level or descent), 20 40 18 observed in the safety window at t_0 . 16 20 14 12 0 10 2/ -1 0 3 4 6 7 8 9 10 11 12 13 *i* 1 8 6 2 0 -2 -4 0-0-13 12 • 11 -8 0 -10 10 -12 0 0 current tactical goal-cell Note: not to scale 0 -0- \cap \cap 0 0 0 0 10 11 12 13 9 -1 0 5 6 7 8

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S_0 : Obstacle Approach (t_1)

Safety Chances Distribution 100 **Safety Window** \rightarrow A fuzzified safety window state at t_1 80 • 11: Commanded bank angle, deg. is shown. The white rectangular in the S6. F14 -45 -37.5 -30 -22.5 -15 -7.5 0 7.5 15 22.5 30 37.5 window is a current tactical goal-cell 24 60 deg. χ^j, % 22 $(\theta_{G2} / \gamma_G) = (0 / -15^\circ)$. Still no threat is 20 40 Commanded flight path angle (climb, level or descent), 18 observed in the safety window. 16 20 14 12 0 10 -1 0 3 6 7 8 9 10 11 12 13 *i* 2/ 4 8 6 4 2 0 -4 13 0 12 -8 11 Ð -10 10 1 \sim current tactical goal-cell Note: not to scale \bigcirc \bigcirc -O $-\bigcirc$ \cap \mathbf{C} $-\bigcirc$ 0 \odot 10 11 12 13 9 -1 0 6 8 5 © 2007 Intelonics Ltd.



S₀: Obstacle Approach (t₂)





S₀: Obstacle Approach (t₃)





S₀: Obstacle Approach (*t*₄**)**





S₀: Obstacle Approach (t₅)





S₀: Obstacle Approach (*t*₆**)**





S₀: Obstacle Approach (*t*₇**)**





course.

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5

S_{\downarrow}: Imminent Collision (t_8)

Safety Window

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ਨੂੰ 🔂 🕂 🤳 🔰 🖉	14	-45	-37.5	-30	-22.5	-15	-7.5	0	7.5	15	22.5	30	37.5	45			
-	24															60	
deç	22														χ ^j , ^o	6	
ent)	20															40	•
lesc	18																
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ngle	6																
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t pal	2																
fligh	0																
ded	-2																
nan	-4																
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9 10 11 12 13

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S_{\downarrow} : Imminent Collision (t_9)





S_{\downarrow}: Imminent Collision (t_{10})





S_{\downarrow}: Imminent Collision (t_{11})





S_{\downarrow}: Imminent Collision (t_{12})





S_{\downarrow}: Imminent Collision (t_{13})





Kazimir Malevich's 'The Black Square' Painting and '9/11'

S ₅ .F ₁₄		• Description of the second state of the se												
		-45	-37.5	-30	-22.5		-7.5				22.5	30	37.5	45
	24													
8														
in i	20													
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9														
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11Bi														
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anim.														
3	-8													
÷	-10													

→ The safety window state just before collision point ($\mathbf{S}_{\downarrow} | t_{13}$), perhaps, helps better understand the meaning of Kazimir Malevich's painting 'The Black Square' - The fatal end is imminent. And there is no chance left to remedy the situation ...



K. Malevich. 'The Black Square' (1913)

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'Last Chance for Recovery' Point $(t_{\uparrow} \equiv t_7)$

Safety Window



current tactical goal-cell

→ However, the 'last chance for recovery' point (t_{\uparrow}) does exist, and it must be assigned to t_{7} . This is marked by the system state when the new 'black' zone (induced by the obstacle) in the safety window first time overlaps with the current tactical goal-cell of the operator's flight control.



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χ^j, %



<u>Legend</u>: **1** – zone of $\Delta\Phi$ -secured non-catastrophic scenarios; **2**, **3** – zones ('islands') of remaining safe/conditionally safe scenarios; **5**, **8** – 'C.G.' locations for left- and right-hand 'islands' of remaining safe/conditionally safe scenarios; **4**, **7** – old (catastrophe-prone) and new (safety restoring) cells of the commanded flight path and bank angles, **6** – required shift of the tactical flight goal-cell in the safety window.



$S_0 \rightarrow S_{\uparrow}$: Self-Preservation Automatic Decision Making at $t_7 \equiv t_{\uparrow}$

Safety Window



→ Based on results of safety 'topology' analysis at t_7 , a self-preservation decision must be made - the current tactical control goal is shifted from the old ('black', collision-prone) cell, $(\theta_{G2}/\gamma_G) = (0/-15^\circ)$, to a new ('green', safe) cell, $(\theta_{G2}/\gamma_G) = (6^\circ/+30^\circ)$, located in the right-hand 'safety island' of the window.





S \uparrow : Collision Avoidance (t_{14})

Safety Window



→ The 'black' zone in the safety window is still expanding (due to vehicle dynamics lag). However, the 'red' zone begins to shrink, and the 'yellow' zone size remains unchanged. The commanded (tactical goal) cell is now located outside the danger and catastrophe-prone zones.



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<u>o 2</u>

01

0

• -1



S \uparrow : Collision Avoidance (t_{15})

Safety Window



100

80

60

40

20

0

-1 0

2

χ^j, %

→ A positive (recovery) safety trend begins to develop. The 'yellow' zone is expanding, and the 'red' zone is shrinking in the safety window.





S \uparrow : Collision Avoidance (t_{16})

Safety Window



→ Positive safety trend remains steady. The 'yellow' zone is expanding, and the 'red' zone is shrinking. The 'black' zone begins to shrink as well ...







S \uparrow : Collision Avoidance (t_{17})

Safety Window



3

2

100

80

60

40

20

0

-1 0

1

χ^j, %

 \rightarrow 'Black' zone induced by the obstacle is about to disappear. Positive safety trend is now irreversible. The 'yellow' zone continues to expand.





S \uparrow : Collision Avoidance (t_{18})

→ The safety window is about to resume its initial Φ₁₁: Commanded bank angle, deg. S6. F14 -37.5 -30 -22.5 -15 -7.5 0 7.5 15 22.5 30 37.5 45 -45 state (vehicle's performance only) as the obstacle 24 (a tower type building) has been safely avoided. $oldsymbol{\Phi}_{ extsf{s}}$: Commanded flight path angle (climb, level or descent), deg 22 20 18 16 14 12 10 8 6 2 0 -2 -4 -6 -8 -10 13 -12 11 10 **Safety Chances Distribution** 100 80 60 χ^j, % 3 b 40 <u>o</u> 2 20 **0**1 0 $\dot{\mathbf{O}}$ **O** 14 15 16 17 18 19 *i* -1 0 5 6 2 3 **•** -1

Safety Window

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S \uparrow : Collision Avoidance (t_{19})

Safety Window



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'Bird's Eye' View Of Dynamic Safety Window Tree: Catastrophic and Recovery Scenarios

*t*₁₃ ¢ • t₁₉ t₁₂ ¢ t_{11} ($O t_{17}$ S_↑ S↓ $t_{10} \, \phi$ t_9 Ot_{15} t_8 *t*₆ 'last chance for *t*₅ recovery' point ('fate switch') *t*₄ *t*₃ \mathbf{S}_0 *t*₂ C t_1 *t*₀ Ç © 2007 Intelonics Ltd.

→ This is a safety window time-history tree. It provides a systematic – 'bird's eye' view level – picture of two alternative scenarios of aircraft flight control in the presence of an urban type obstacle. Such obstacles can be a part of a multi-factor flight situation domain-'neighborhood' of the current situation.

Legend:

Scenario segments:

- \mathbf{S}_0 obstacle approach
- S_{\downarrow} imminent collision
- S_{\uparrow} AI based collision avoidance

Scenario time lines:

 $\begin{array}{l} \{t_0, \ t_1, \ ..., \ t_7\} - \mathbf{S}_0 \\ \{t_8, \ ..., \ t_{13}\} \ - \mathbf{S}_\downarrow \\ \{t_{14}, \ ..., \ t_{19}\} \ - \mathbf{S}_\uparrow \end{array}$

Key time instants:

- t_7 'last chance for recovery'
- t_{13} 'just before impact'
- t_{19} 'safety restoration complete'

Safety Chances Distribution Time-History for Two Control Tactics





(2) AI based self-preservation control



Legend: A, B, ..., L - characteristic states of the aircraft safety dynamics; χ^j – flight safety chances at ξ^j level, $j \in \{I, II-a, II-b, III, IV, V\}$; t_i – time instants, $i \in \{-1, 0, 1, ..., 13\} \lor i \in \{-1, 0, 1, ..., 7, 14, 15, ..., 19\}$.

→ Characteristic states {A, B, C, ..., L} of the vehicle's safety dynamics and their recognition criteria are expedient to use in the automatic or manual recovery decision-making process in emergency situations under uncertainty. In accordance with the self-preservation imperative for a civil aircraft, flight control authority in a life-threatening situation must be dynamically assigned/transferred to a most competent

agent.

Conclusions



- Generalized knowledge-centered methodology has been developed for UAV flight safety prediction and protection in multifactor situations near operational constraints.
- 2. Method's advantages are: use of integrated conceptual framework, simple realtime calculations, open memory-based knowledge system, situation-independent decision-making algorithm, exploration of situation 'what-if neighborhood' tree for short-term flight path probing, use of 'bird's eye' view 'topology maps' for flight safety status monitoring and automatic recovery in emergencies.
- **3.** However, prerequisites for successful implementation of developed methodology are:
 - → availability of vehicle's validated 'parametric definition' database, and
 - onboard integrated sensor suit capable of detecting potentially dangerous physical/ virtual obstacles inside vehicle's 'safety ellipsoid/cone'.
- **4.** Potential application areas are as follows:
 - → design of affordable, yet expert pilot level AI safety protection systems based on self-preservation imperative for unmanned/ manned air vehicles to prevent key accident/ incident scenarios such as LOC, CFIT, 'pilot error', hardware failure, mid-air collision, and '9/11'
 - design of adaptive mission control and autonomous collision avoidance systems (integrated with C.Reynolds swarming model, ethology principles, etc.) for heterogeneous multivehicle clusters and free-flight operations.



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Questions, please



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